

Large scale flows in natural and mixed convection

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- Convective turbulence in closed volumes is associated with large-scale circulations of the flow (LSC).
- LSC and the amount of heat are determined by proper orthogonal decomposition (POD) of the field.
- The most energetic POD modes give insight into the dynamic dominance of coherent flow and temperature patterns: How much do they contribute to the global momentum & heat transfer?
- Influence of the geometry and inflow conditions needs to be clearly understood.

Numerical model

$$u_t + u \cdot \nabla u + \nabla p = Gr^{-1/2} \nabla^2 u + T e_z$$

$$\nabla \cdot u = 0$$

$$T_t + u \cdot \nabla T = Gr^{-1/2} Pr^{-1} \nabla^2 T$$

with Parameters:

$$Gr = \alpha g D^3 \Delta T / \nu^2, \quad Pr = \nu / \kappa = 0.7, \quad Ra = Gr Pr = 10^7 - 10^9$$

POD ANALYSIS:

- For the Fluctuations $\mathbf{v} = (u_1, u_2, u_3, \theta)$, we seek Φ , such that $\frac{\langle |(\mathbf{v}, \Phi)|^2 \rangle}{\|\Phi\|^2}$ is maximized

- The SOLUTION is given by

$$K \Phi = \iint K(x, x') \Phi(x') d x'^3 = \lambda \Phi(x)$$

$$K_{ij}(x, x') = \langle v_i(x) v_j^*(x') \rangle$$

- THE METHOD OF SNAPSHOTS:

$$\Phi(x) = \sum_{n=1}^M \alpha_n v^{(n)}(x) \Rightarrow \left(\frac{1}{M} \iiint_V \sum_{i=1}^4 v_i^{*(m)} v_i^{(n)} d x^3 \right) \alpha(n) = \lambda \alpha(m)$$

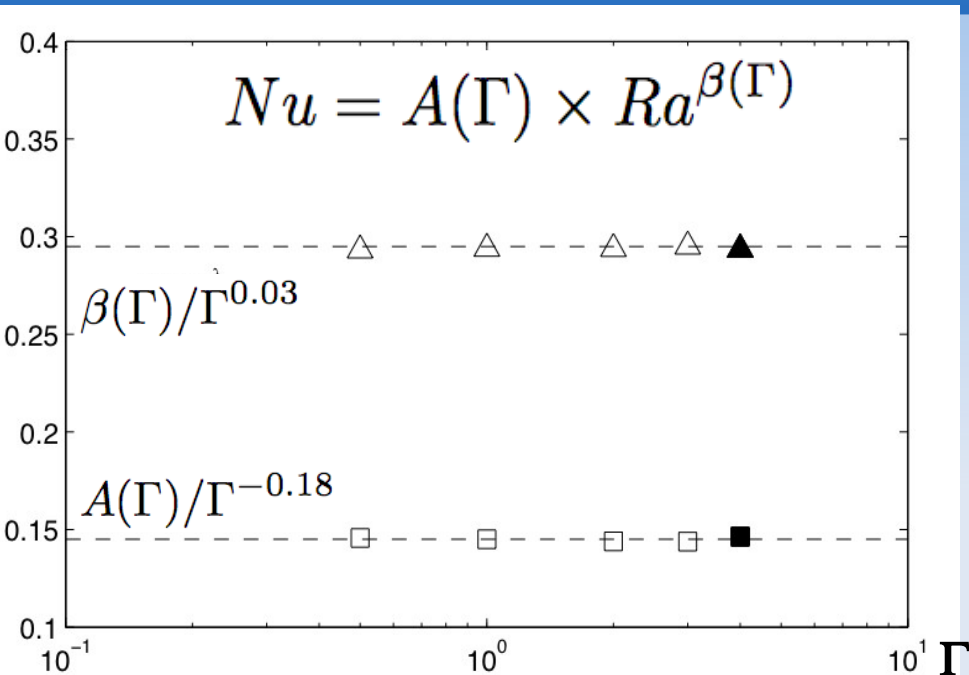
- TIME DEPENDENT COEFFICIENTS:

$$v(x, t) = \sum_n a_n(t) \Phi^{(n)}(x)$$

$$a_n(t) = (\Phi^{(n)}(x), v(x, t))$$

$$a_n(t) = \iiint_V \Phi_i^{*(n)}(x) v_i(x, t) d x^3$$

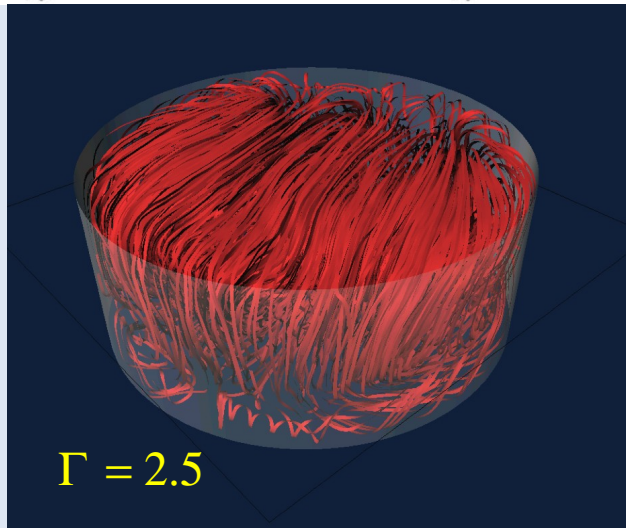
Previous work: geometry dependence



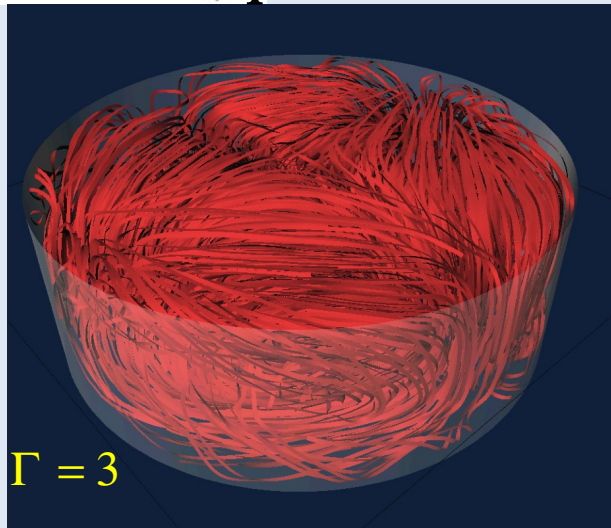
**Bailon-Cuba, Emran & Schumacher,
J. Fluid Mech., submitted (2009)**



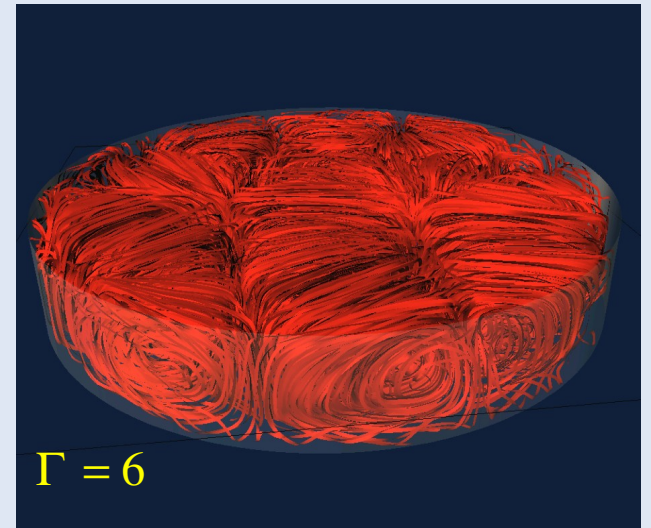
**Niemela & Sreenivasan,
J. Fluid Mech. (2006)**



$\Gamma = 2.5$



$\Gamma = 3$



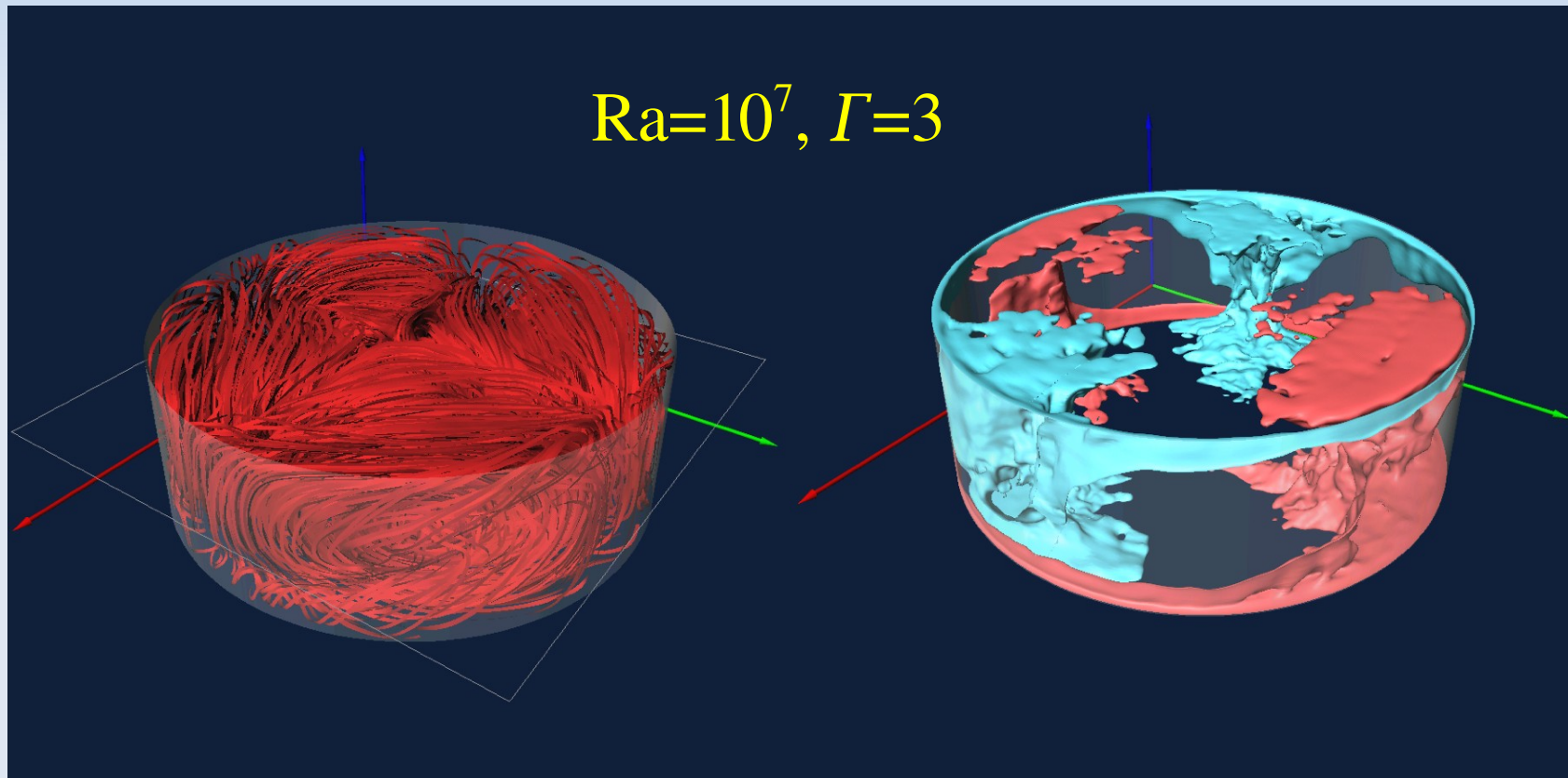
$\Gamma = 6$

**Systematic dependence of LSC and global heat transport with the
aspect ratio Γ ($Ra=10^7$)**

POD analysis of free convection

Boundary Conditions:

- No-slip at horizontal plates $u=0$, & at a fixed temperature
- Side walls adiabatic & no-slip, $u=0$ and $\partial T/\partial n=0$



Streamlines of first flow
mode

Iso-surfaces of first
temperature mode

Turbulent heat transfer

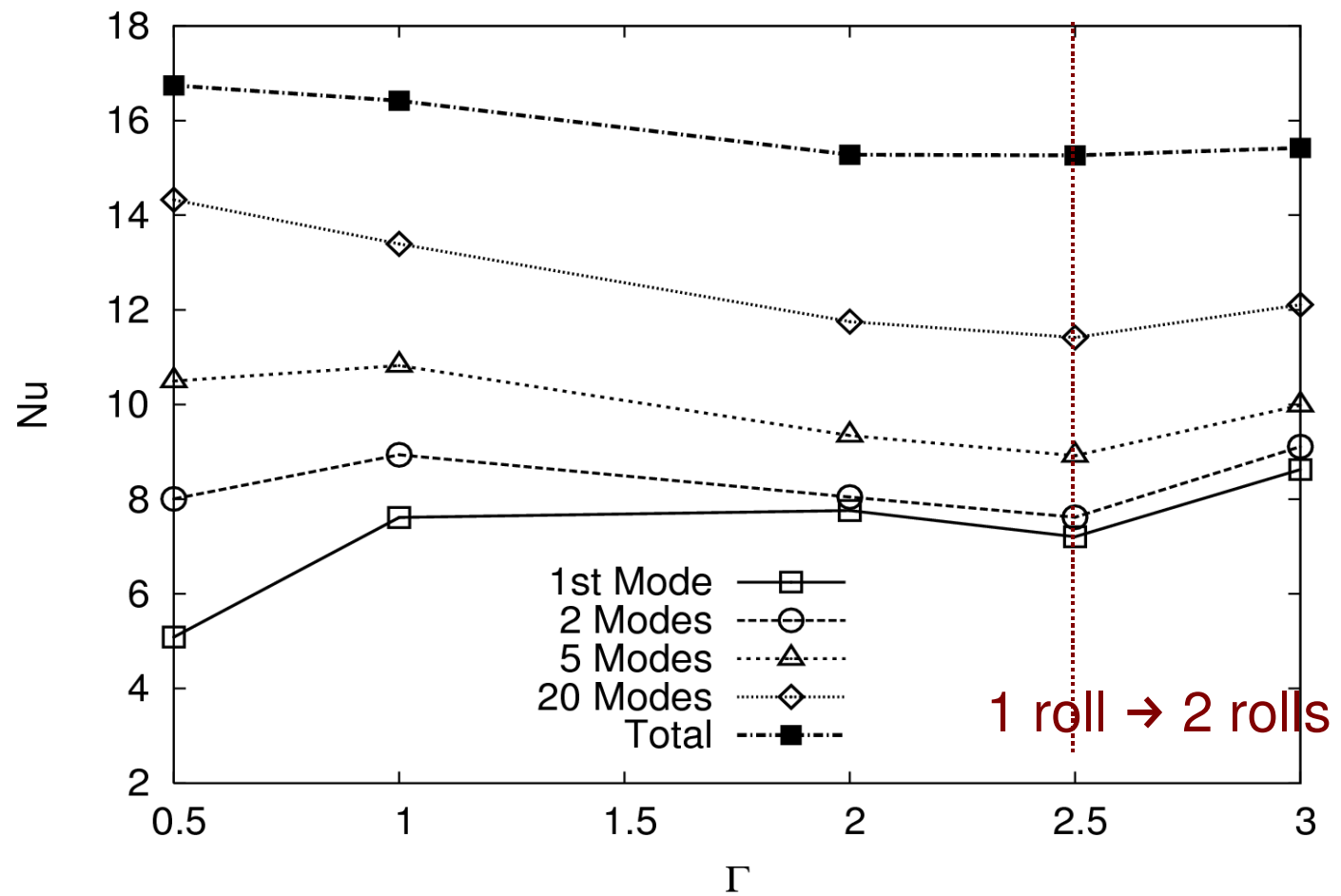
- Nusselt Number

$$Nu = 1 + \frac{H}{\kappa \Delta T} \sum_{m,n=1}^M \langle a_m(t) \Phi_3^{(m)}(\mathbf{x}) a_n(t) \Phi_4^{(n)}(\mathbf{x}) \rangle_{V,t}$$

- $a_m(t)$ correspond to the projection of turbulent flow field at time t to mode $\phi^{(m)}(\mathbf{x})$

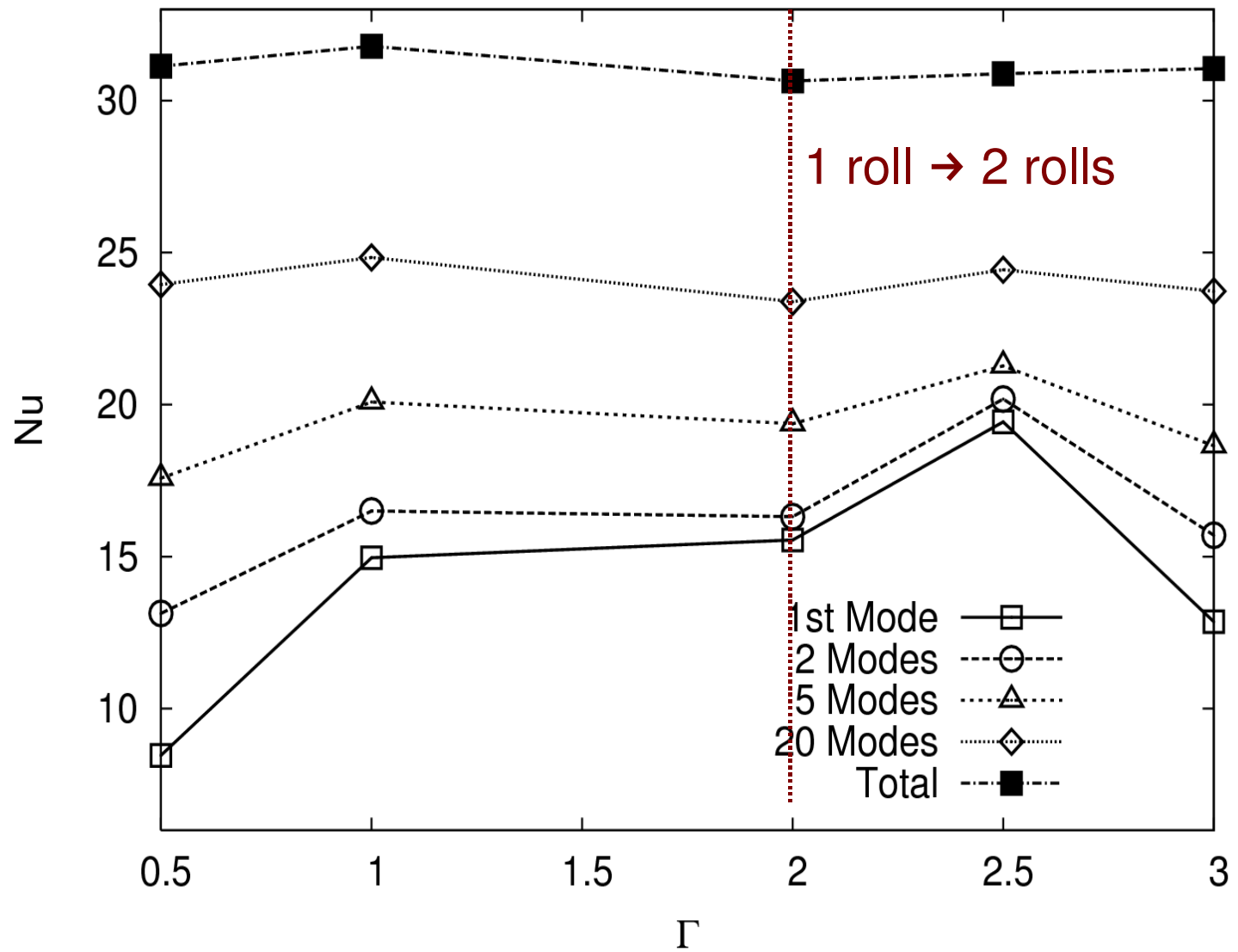
($Ra = 10^7$)

Bailon-Cuba,
Emram &
Schumacher, JFM,
submitted (2009).

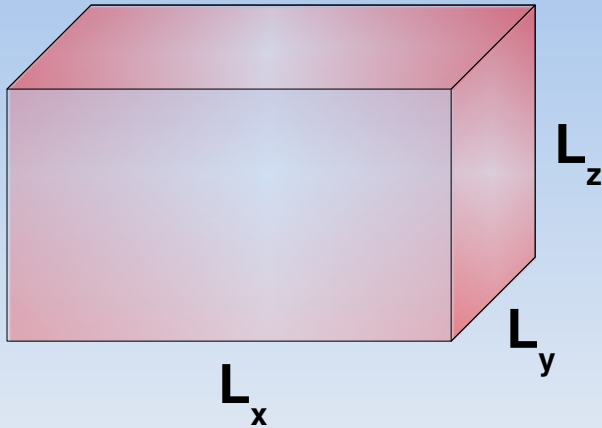


Turbulent heat transfer²

($Ra = 10^8$)
Bailon-Cuba,
Emram &
Schumacher, JFM,
submitted (2009).



Rectangular geometry: canonical problem



- $\Gamma = L_x/L_z = L_y/L_z = 2$

- **BC's: Periodic in x & y**

At $z=0$, L_z (free slip): $w = T = \partial u / \partial z = \partial v / \partial z = 0$

- $Ra = 9.9 \times 10^5$, 60 Snapshots: $128 \times 128 \times 65$

- Fourier transform in x,y and Fourier Kernel

$$F_j(n_x, n_y; z) = \sum_x \sum_y v_j(x, y, z) e^{2\pi i(n_x x/L_x + n_y y/L_y)}$$

$$\kappa_{ij}(n_x, n_y; z, z') = \langle F_i(n_x, n_y; z) F_j^*(n_x, n_y; z') \rangle$$

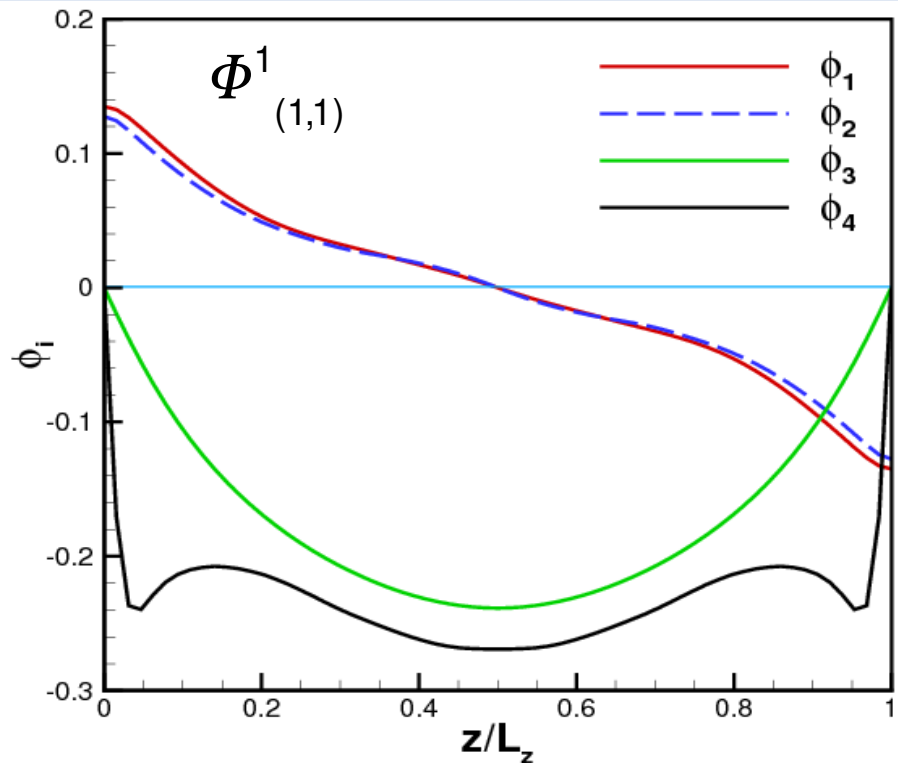
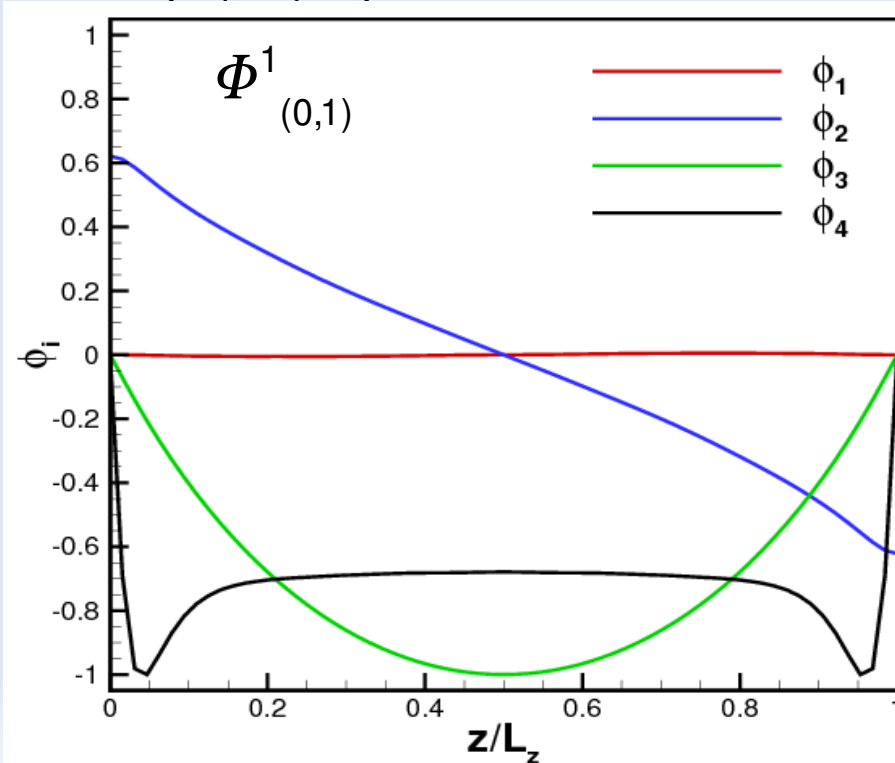
- Method of Snapshots

$$\left(\int_0^{L_z} \frac{1}{M} \sum_{i=1}^4 F_i^{*(m)}(n_x, n_y; z') F_i^{(n)}(n_x, n_y; z') dz' \right) \alpha_n(n_x, n_y) = \lambda(n_x, n_y) \alpha_m(n_x, n_y)$$

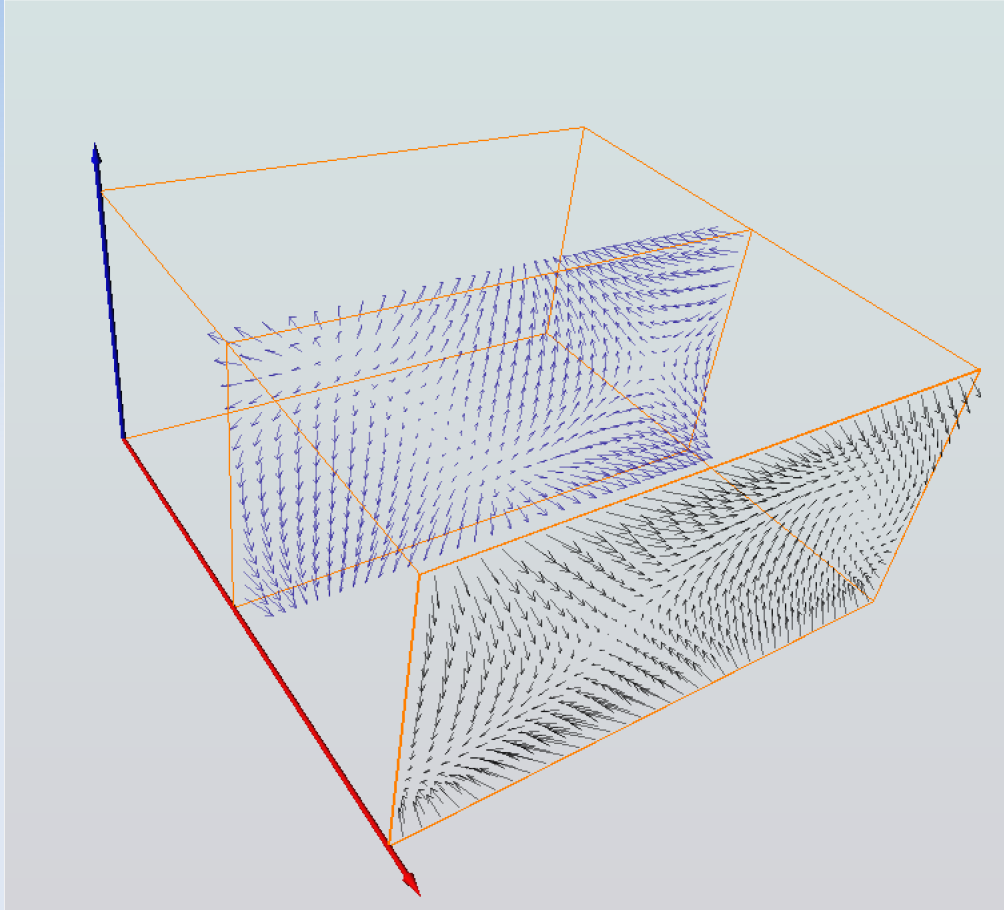
Results

k	(n_x, n_y, n)	λ_{n_x, n_y}^n	Degeneracy*	% Energy
1	(0, 1, 1)	1.4130	4	30.078
2	(0, 1, 2)	0.1846	4	3.929
3	(0, 1, 3)	0.1694	4	3.605
4	(1, 1, 1)	0.1678	4	3.571
5	(0, 2, 1)	0.1303	4	2.774
6	(1, 2, 1)	0.0583	8	2.482
7	(0, 2, 2)	0.1077	4	2.293

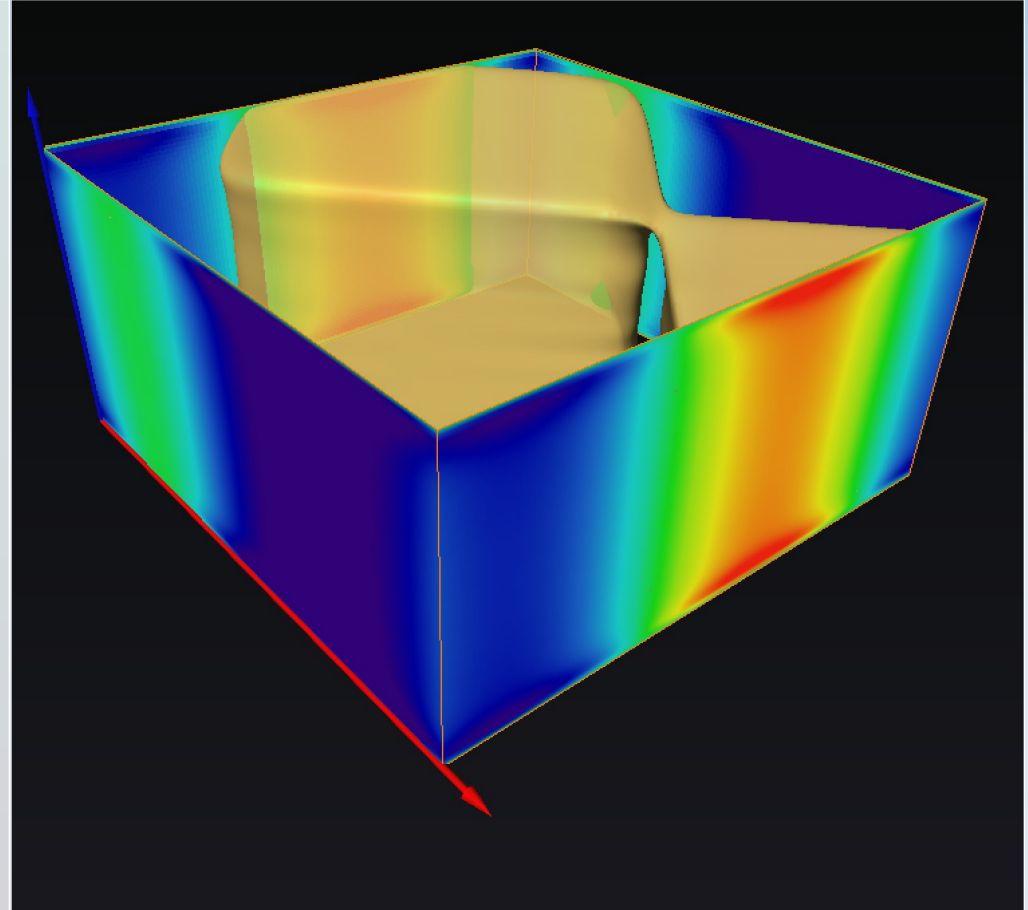
* Due to
The 16
discrete
symmetries:
reflections &
rotations



3D-structure of POD modes

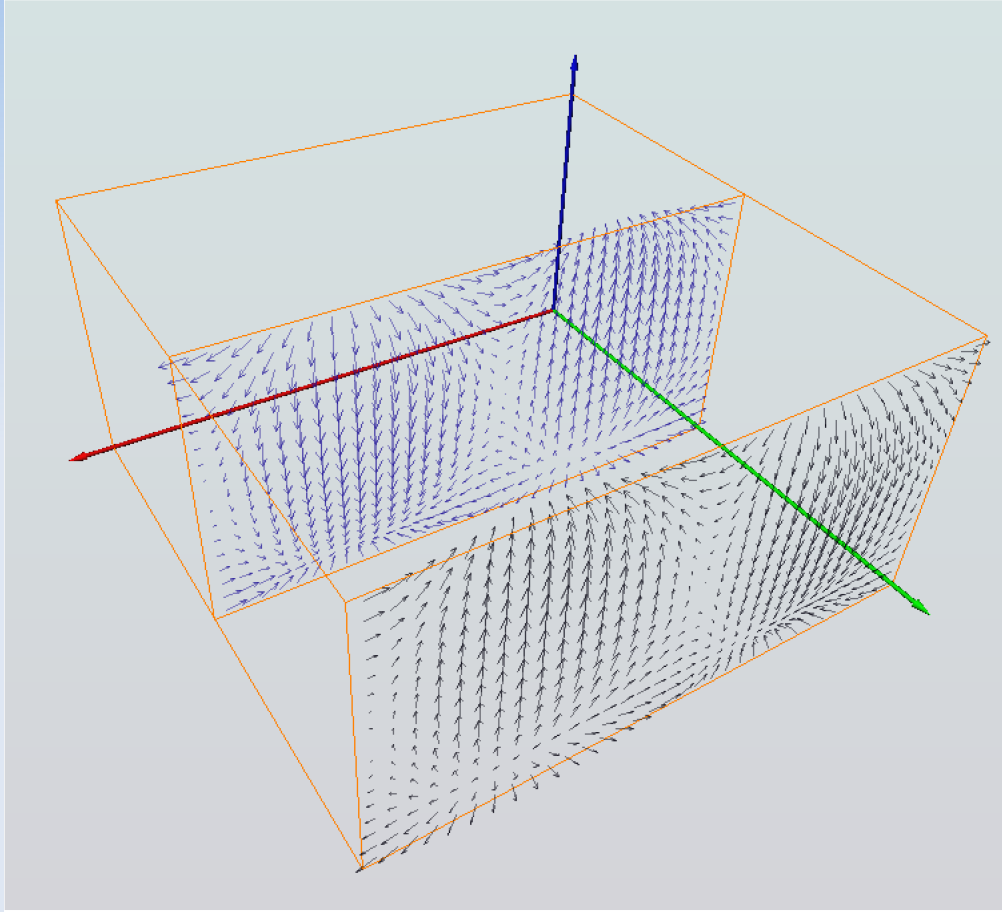


$\Phi_{(0,1)}^1$ - Velocity

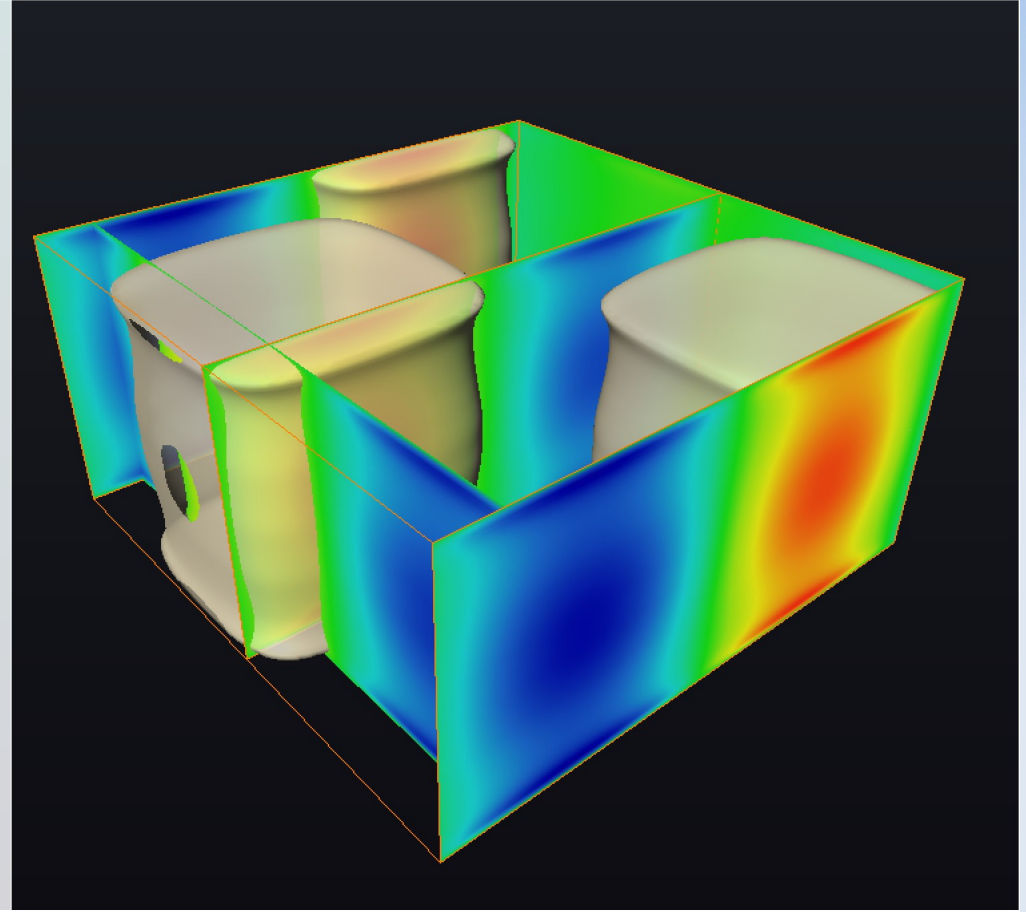


$\Phi_{4(0,1)}^1$ - Temperature

3D-structure of POD modes²



$\Phi^1_{(1,1)}$ - Velocity



$\Phi^1_{4(1,1)}$ - Temperature

Mixed convection

We consider a complex rectangular setting which mimics indoor ventilation problems as present in a passenger cabin of an airplane (or/and concert hall)



Air plane cabin



Concert hall

Rectangular geometry

Boundary Conditions:

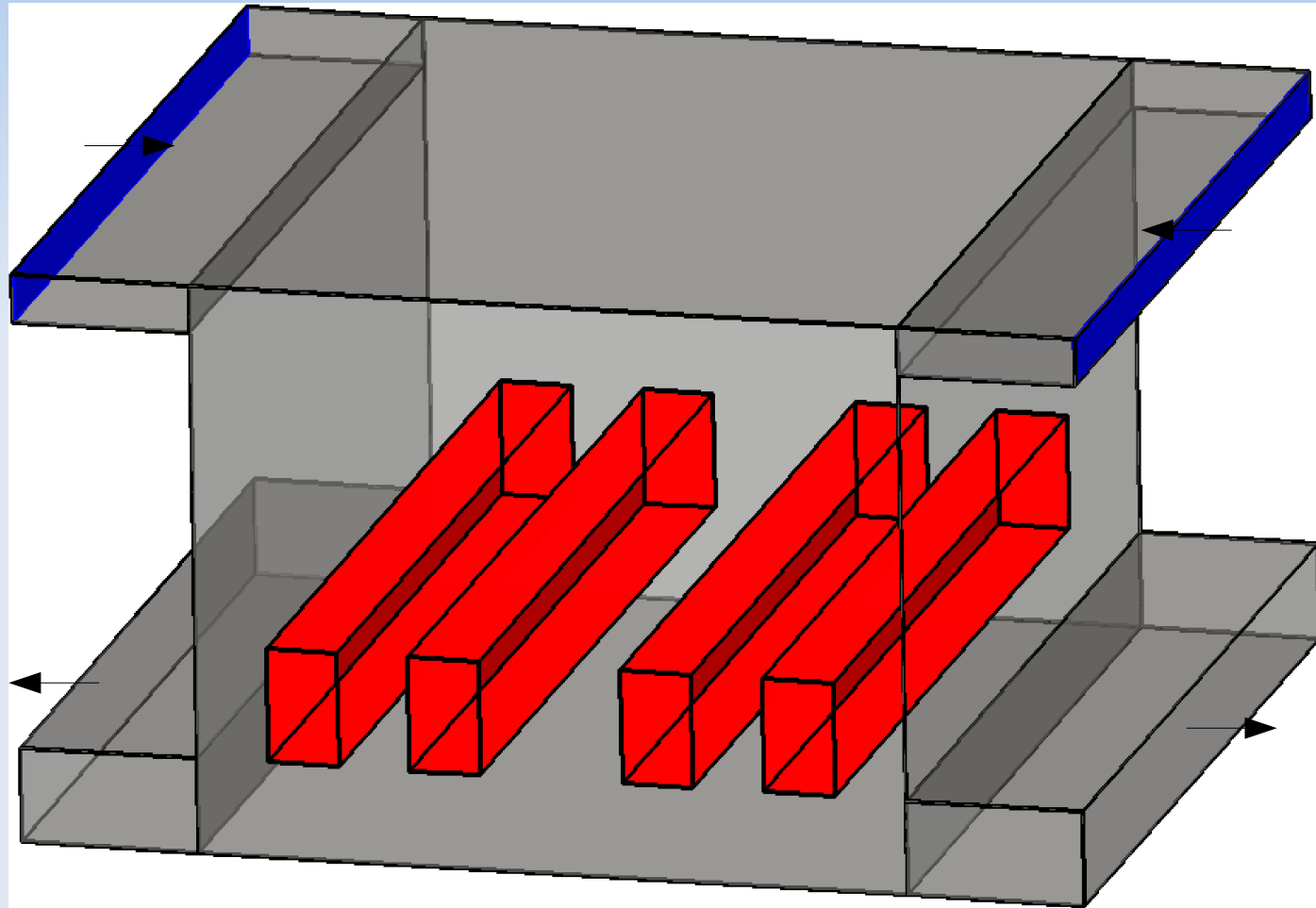
$$\left. \begin{array}{l} T = -0.5 \\ u = 5 \text{ m/s} \end{array} \right\} \text{ at the Inlet}$$

$$T = 0.5 \text{ at the Obstacles}$$

$$\partial T / \partial \mathbf{n} = 0 \text{ Boundaries of Outlet Ducts \& Walls}$$

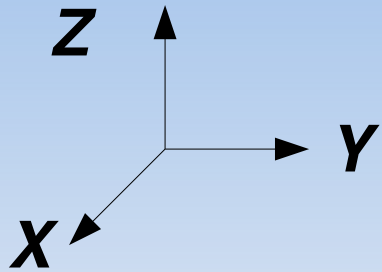
$$\partial u / \partial \mathbf{n} = 0 \text{ Boundaries of Outlet Ducts}$$

$$u = 0 \text{ At the walls}$$



Domain for the DNS by Shishkina & Wagner.

Rectangular geometry: symmetries



$\{ I, R^2, X, Y \}$ Isomorphic to Abstract Group D2

whose elements are the rotation by 180°

$$R^2(x, y, z, u, v, w, \theta) = (-x, -y, z, -u, -v, w, \theta)$$

and the reflections in x & y

$$X(x, y, z, u, v, w, \theta) = (-x, y, z, -u, v, w, \theta)$$

$$Y(x, y, z, u, v, w, \theta) = (x, -y, z, u, -v, w, \theta)$$

The set of snapshots is extended Four Times:

$$\left(\frac{1}{4M} \iiint_V \sum_{i=1}^4 v_i^{*(m)} v_i^{(n)} d\mathbf{x}^3 \right) \alpha(n) = \lambda \alpha(m)$$

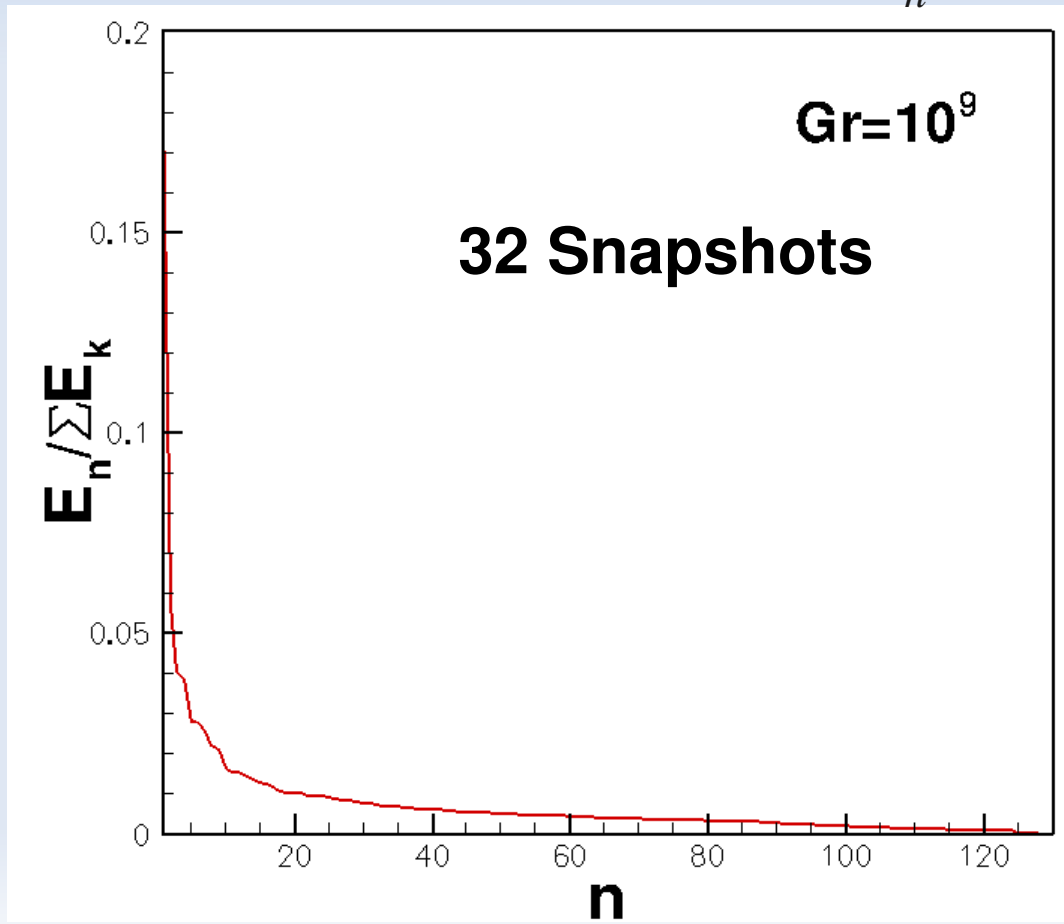
$$m, n = 1, 2, 3, \dots, 4M$$

Energy content of the POD modes

$$u'_i(\mathbf{x}, t) = u_i(\mathbf{x}, t) - \bar{U}(x, z) = \sum_n a_n(t) \Phi_i^{(n)}(\mathbf{x})$$

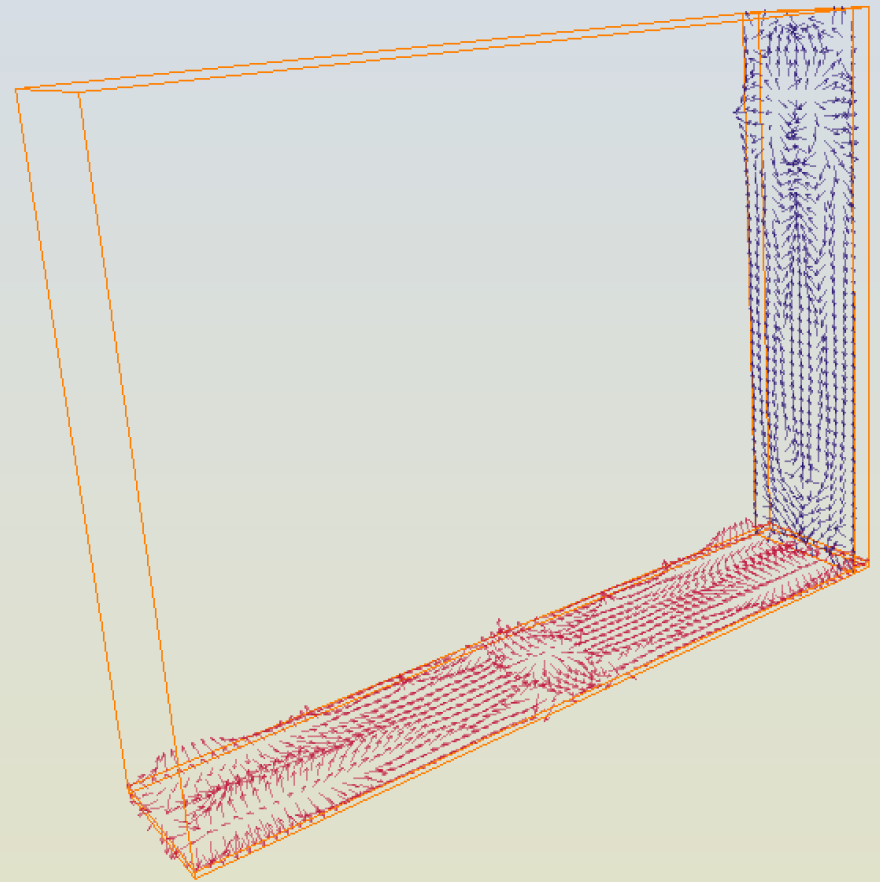
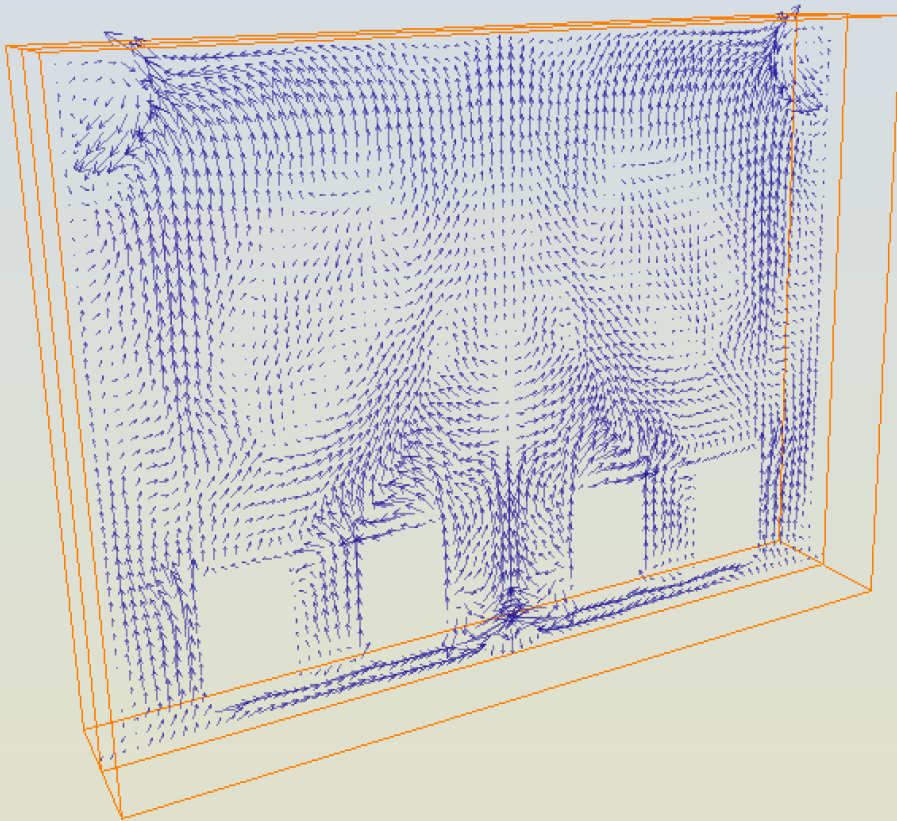
$$i=1,2,3$$

$$\theta(\mathbf{x}, t) = T(\mathbf{x}, t) - \bar{T}(x, z) = \sum_n a_n(t) \Phi_4^{(n)}(\mathbf{x})$$



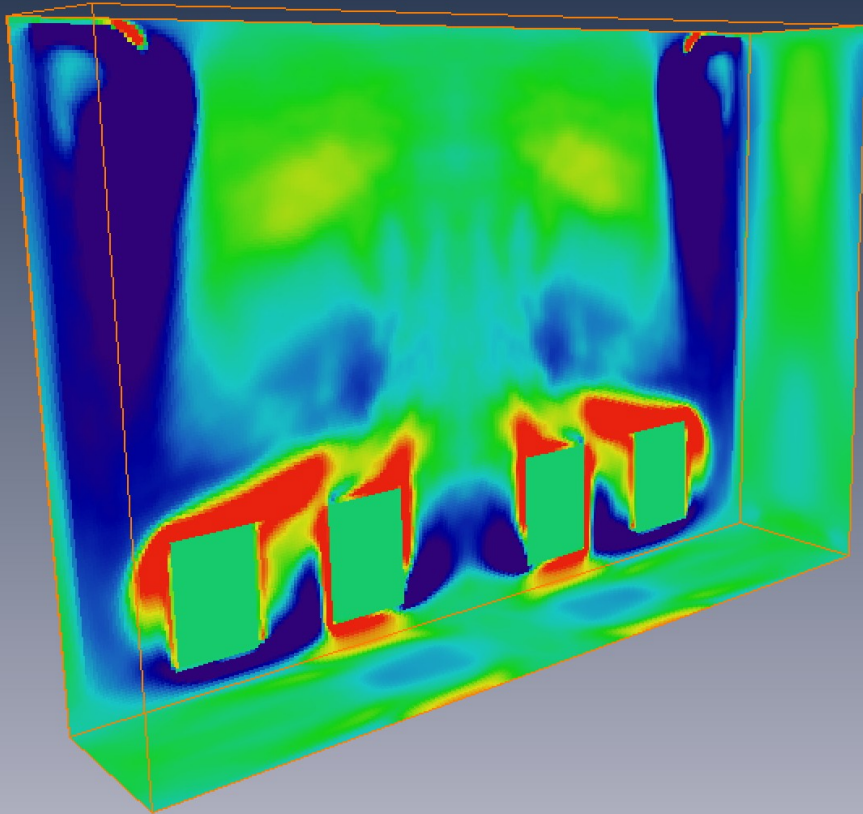
Mode	% Energy
1	18.27
2	5.72
3	4.06
4	3.88
5	2.80
6	2.77
7	2.59
8	2.20
9	2.13
10	1.66
11	1.54
12	1.53

First mode: velocity field

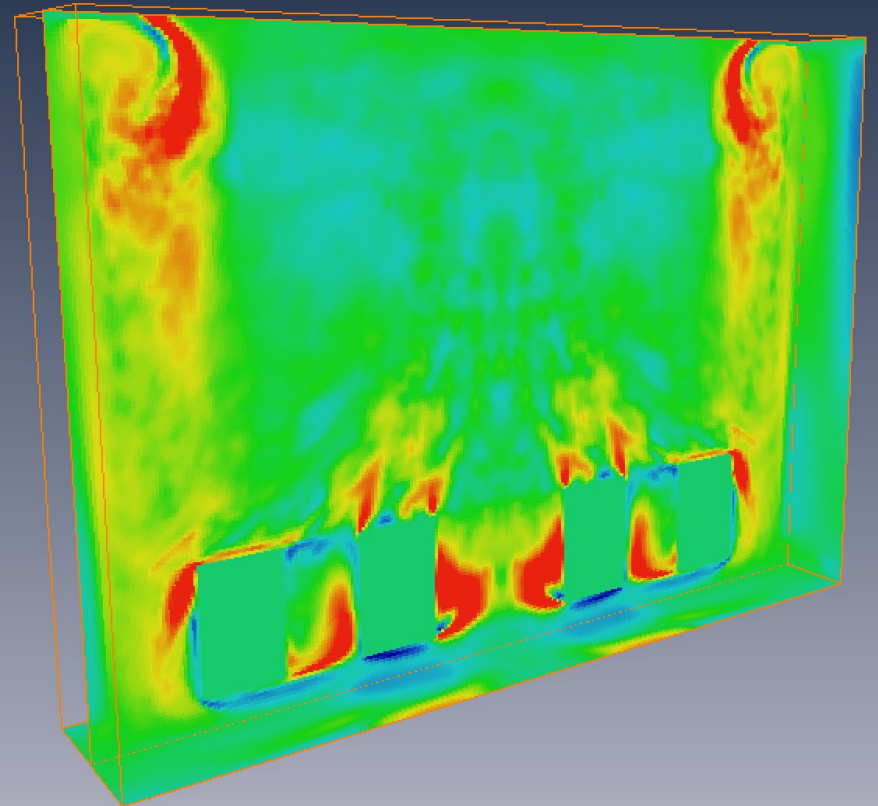


32 Snapshots (DNS by Shishkina & Wagner)

First mode: temperature field



At the wall interface



At the middle plane

32 Snapshots (DNS by Shishkina & Wagner)

Nusselt number-POD modes dependence

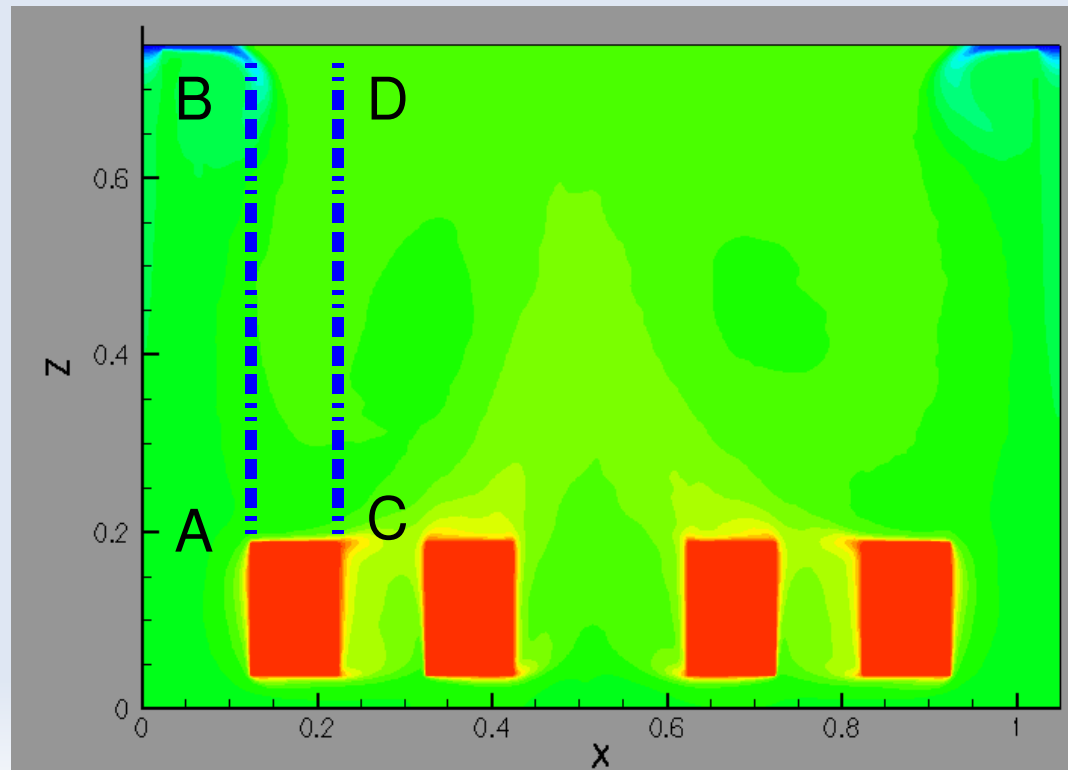
- From the energy equation $T_t + u \cdot \nabla T = \nabla^2 T$

- Taking the ensemble average & along “y”

$$\frac{\partial \langle uT \rangle}{\partial x} + \frac{\partial \langle wT \rangle}{\partial z} = \frac{\partial^2 \langle T \rangle}{\partial x^2} + \frac{\partial^2 \langle T \rangle}{\partial z^2}$$

- And after horizontal averaging inside the sub-volume ABCD:

$$\frac{d \langle wT \rangle}{dz} = \frac{d^2 \langle T \rangle}{dz^2}$$



Nusselt number-POD modes dependence²

$$\left. \frac{d \langle T \rangle_{y,t}}{dz} \right|_{AC} = \boxed{\frac{d \langle T \rangle_{y,\Delta x,t}}{dz}} - \langle wT \rangle_{\Delta V,t}$$

**Known from Average
Temperature Field**

NEXT STEPS

- To complete the study of the Nu -POD Modes dependence for $Ra=10^8$ & 10^9
- To identify the large scale flow patterns for these geometries.
- To start a coarse graining procedure in order to determine the integral heat & momentum fluxes.